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THESIS

COMPOZITE IN-SITU CU MATRICE DE ALUMINIU ȘI PARTICULE DE ZrB₂

IN-SITU COMPOSITES WITH ALUMINUM MATRIX AND ZrB_2 PARTICLES

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Keywords

Composites Metal Matrix Composites (MMC) Particles Fibers In Situ Processing Reinforcement Aluminium metal matrix composites (AMCs) Zirconium diboride (ZrB₂) Ceramic Reinforcement Casting Aluminium

Introduction

Metal composite materials have found application in many areas of daily life for quite some time. Often it is not realized that the application makes use of composite materials. These materials are produced in situ from the conventional production and processing of metals [1].

Materials like cast iron with graphite or steel with a high carbide content, as well as tungsten carbides, consisting of carbides and metallic binders, also belong to this group of composite materials. For many researchers the term metal matrix composites is often equated with the term light metal matrix composites (MMCs). [1]

Substantial progress in the development of light metal matrix composites has been achieved in recent decades, so that they could be introduced into the most important applications. In traffic engineering, especially in the automotive industry, MMCs have been used commercially in fibre reinforced pistons and aluminium crank cases with strengthened cylinder surfaces as well as particle-strengthened brake disks. [1]

These innovative materials open up unlimited possibilities for modern material science and development; the characteristics of MMCs can be designed into the material, custom-made, dependent on the application. From this potential, metal matrix composites fulfill all the desired conceptions of the designer. This material group becomes interesting for use as constructional and functional materials, if the property profile of conventional materials either does not reach the increased standards of specific demands or is the solution of the problem. However, the technology of MMCs is in competition with other modern material technologies, for example powder metallurgy. The advantages of the composite materials are only realized when there is a reasonable cost – performance relationship in the component production. The use of a composite material is obligatory if a special property profile can only be achieved by application of these materials. [1]

Titanium, aluminium and magnesium alloys are the most popular matrix metals presently in vogue, which are particularly suitable for automobile, defence, structural and aircraft applications [1]. In the last three decades, metal matrix composites (MMCs) have the potential to replace the conventional materials in several fields of applications like transportation, military, marine as well as in various advanced engineering industries [2].

Aluminium matrix composites (AMCs) are being considered as a group of advanced materials for their lightweight, low thermal expansion coefficient, outstanding wear resistance properties and good mechanical properties [3].

Extensively employed fabrication methods for aluminium matrix composites involve stir casting, compo casting, vacuum casting, powder metallurgy, centrifugal casting, insitu casting and squeeze casting [4-5]. Among those available process, in situ method is most economical and is always preferred. In situ formed particles reveal strong interfacial bonding with the matrix. In situ method overcomes the limitations of stir casting process such as improper wetting of reinforcement particles and density dependence of particles and its associated problems like sinking and floating of particles [6]. In situ ceramic particles, such as Al₂O₃, TiB₂, AlN, TiC, B₄C and ZrB₂ have been widely used as reinforcements in aluminium-based composites [7].

The particular attributes of aluminium composites are a combination of high specific stiffness, good fatigue properties, and the potential for relatively low-cost conventional processing. It is also possible to tailor the mechanical and thermal properties of these materials to meet the requirements of a specific application. To do this there are a number of variables which need to be considered, which include the type and level of reinforcement, the choice of matrix alloy, and the

composite processing route. All these factors are inter-related and should not be considered in isolation when developing a new material [8].

Aluminium composites have been under development for many years during which time a vast number of different types of reinforcement have been attempted with varying degrees of success [9]. These include continuous fibres, both monofilament and multifilament, short fibres, whiskers and particulates [10]. Many different matrices have been tried over the years and these have a bearing on some of the properties that can be achieved in the composite. Corrosion resistance, strength levels, toughness etc. are all strongly influenced by the matrix alloy [11]. Generally standard engineering alloys are used but in a slightly modified form to accept the selected reinforcement.

The type of reinforcement also influences the method of manufacture, continuous monofilament needs to be handled in a different way to particulate or even short fibre reinforcement. The aluminium composites currently under consideration, by the auto industry, for application in gas turbine engines that are particulate reinforced [8].

Even with this restriction a number of processing routes may be employed, and secondary processing may be applied to further tailor the material properties to meet a particular component requirement. The great advantage of particulate reinforcement, in terms of processing, is that conventional metal manufacturing methods and machining techniques can be used. This improves the economics of the case for the use of aluminium metal matrix composites relative to that of other composites, which have, traditionally, been expensive and very labour intensive [8].

CHAPTER 1. CLASSIFICATION OF METAL MATRIX COMPOSITES

1.1. Types of metal matrix composites

Metal matrix composites can be classified in various ways. One classification is the consideration of type and contribution of reinforcement components in particle-, layer-, fiber and penetration composite materials (see Fig. 1.1) [1]. Fiber composite materials can be further classified into continuous fiber composite materials (multi and monofilament) and short fibers or, rather, whisker composite materials, see Fig. 1.2. [1]



Figure 1.2. Schematic representation of three forms of metal matrix composites [1]

1.2. Reinforcement phases

Reinforcements for metal matrix composites have a manifold demand profile, which is determined by production and processing and by the matrix system of the composite material. The following demands are generally applicable [4]: [1]

• low density,

• mechanical compatibility (a thermal expansion coefficient which is low but adapted to the matrix),

- chemical compatibility,
- thermal stability,
- high Young's modulus,
- high compression and tensile strength,
- good processability,
- economic efficiency. [1]

1.3. Particles, fibers and short fibers used for reinforcing metallic materials

The availability as well as the demand for reinforcing compounds for metal matrix composites is very extensive. Their selection depends on the condition of the matrix, the type of processing of the composite material and the demands on the material (temperature, corrosion, stress etc.). [1]

These demands can be almost exclusively fulfilled by nonmetal inorganic reinforcement components. Ceramic particles, or rather fibers or carbon fibers, are used for metal reinforcement. An application area of metal fibers is that of functional materials (for example for contacts, conductors and superconductors). However, their application in the structural area mainly fails because of the high density. Organic fibers cannot be employed because of their low Young's modulus, processing problems, poor thermal stability and poor compatibility [12].

Reinforcement materials for metal matrix composites can be produced in the form of continuous fibers, short fibers, whiskers, or particles. The parameter that allows us to distinguish between these different forms of reinforcements is called the aspect ratio. Aspect ratio is nothing but the ratio of length to diameter (or thickness) of the fiber, particle, or whisker [13].

Thus, continuous fibers have an aspect ratio approaching infinity while perfectly equiaxed particles have an aspect ratio of around one.

Ceramic reinforcements combine high strength and elastic modulus with high temperature capability [13].

Continuous ceramic fibers are also, however, more expensive than ceramic particulate reinforcements [13].

Considering economic criteria, the use of discontinuous reinforcement, like particles or short fibers, appears most favorable [1].

CHAPTER 2. The study of the specialized literature

Numerous studies on the elaboration of aluminum matrix composite materials reinforced with ZrB_2 , $ZrAl_3$ particles or containing both compounds, hybrid composites, are published in the specialized literature ($ZrB_2 + ZrAl_3$) [33 – 52].

Alloys A356 [33, 43, 46], AA6061 [35, 51, 52], AA2024 [36, 40], AA5052 [38, 39, 48], AA2618 [41], AA2014 [42] can be used as matrix material, AA6351 [44], AA7075 [47], AA6061 [47], A380 [50] or metallic aluminum: Al 99.7% powder [34], Al of purity over 98% [37, 45, 53 - 65].

By reactions between aluminum alloys and Zr-containing elements (eg K_2ZrF_6) and boroncontaining elements (eg KBF₄) [34, 37 - 48] or using Al-B and / or Al-Zr pre-alloys [33, 49], various concentrations of reinforcing elements were obtained, as final reaction products, intermediates or combinations thereof.

In addition to aluminothermic reactions, other methods were used such as borothermic and carbothermic reduction, mechanochemical treatment, CVD, sol-gel, thrombolysis of ZrB_2 -containing gas, magnetochemical process, etc. [53 - 65].

The main diagram studied was that of Al - K_2ZrF_6 - KBF₄, but studies were also presented on the binary diagrams KF-NaF, KF-KCl, KCl-NaCl [53 - 65], following the melting temperatures at various concentrations, so that the salts are in the liquid state for the best possible conditions for the development of the formation reactions of the reinforcing compounds.

A very important parameter when obtaining metal matrix composites reinforced with Al₃Zr, ZrB₂ etc. was the working temperature. Thus, from the study of the literature it results that we must have temperatures higher than 700°C [5]: 850°C [33, 42, 44, 47, 49], 860°C [39], 870°C [36, 40], 885°C [38, 48], 900°C [44]. In the study [44], the authors performed the analysis of the compound ZrAl₃ and its morphology at 850°C, 900°C, 950°C and 1000°C.

Another parameter that has an influence both on the process of obtaining composite materials and on the quantity and dimensions of the reaction products, is the stirring time of the mixture formed by the liquid matrix material and the salts used for the addition of B and Zr. Stirring times varied with values ranging from 10 minutes [49], 20 minutes [42], 30 minutes [44, 48], 40 minutes [47], exceeding one hour or reaching up to 4 hours [34].

As a result of the reactions, reinforcing particles with different morphologies were obtained, with dimensions starting from 15 nm [36, 39, 40] and reaching 100 μ m [33, 46, 50, 52].

Samples from the composite materials obtained were analyzed by optical microscopy - MO [35, 36, 40, 42, 44, 47 - 50] and SEM and TEM electron microscopy [33 - 40, 43 - 52], as well as X-ray diffraction (XRD), as can be seen in the following figures (2.1 - 2.5).



Figure 2.1. Optical microscopy of AA2014 (a) alloy, AA2014 / 4% ZrB₂ (b) composite AA2014 / 8% ZrB₂ (c) composite, and XRD for AA2014 / ZrB₂ (d) composite [42]



Figure 2.2. SEM microstructures of particles extracted from the composite material, obtained at 600°C (a) and 800°C (b), reaction time 2 hours [34]



Figure 2.3. SEM microstructures of particles extracted from the composite material obtained at 800oC, reaction time 1 hour (a) and 2 hours (b) [34]



Figure 2.4. SEM microstructures (a, b) and optical microstructure of composite AA6061 / Al_3Zr [35]



Figura 2.5. Imagini SEM ale compozitelor Al/Al₃Zr + ZrB₂ obținute la diferite temperaturi: (a) 1123 K; (b) 1173 K; (c) 1223 K și (d) 1273 K [45]

Other techniques for highlighting the structures and compounds formed are: XRF [34], XRD and EDAX [35, 36, 37, 39, 40, 42 - 52], DTA-TG [37], TEM [33, 36, 39, 40], HRTEM [34, 36]. The properties of the obtained materials were subjected to tests to determine the physical-mechanical properties (hardness, wear resistance, mechanical strength, elongation, expansion) [33, 38, 39, 40, 42 - 50], to determine the corrosion potential, the currents of corrosion, linear polarization resistance, etc. [41].



Figure 2.6. TEM image for ZrB₂ compound, obtained at 800oC, reaction time 2 hours [34]



Figure 2.7. TEM image (s) with highlighting reinforcement elements and HRTEM image (f) of the shaded area with white square (s) [36]



Figure 2.8. Thermogravimetric analysis for composite AA5052 / ZrB₂ [36]



Figure 2.9. Determined mechanical properties for AA2024 / ZrB₂ composite [40]

CHAPTER 3. Studies and research on the thermodynamics of in situ processes of obtaining Al matrix composites reinforced with ZrB₂ particles

Studies and research on the in-situ production of aluminum matrix composites and reinforcing particles in the form of boron are presented in the literature. However, there is no unitary view of the thermodynamics of the interaction processes of aluminum alloys introduced into the melt with KBF₄ and K₂ZrF₆ salts at high temperatures. The addition of salts in molten aluminum, at 890°C, generates the intermetallic compounds Al₃Zr and AlB₂ in the first phase and, after the completion of the reaction, the compound ZrB₂. In order to clarify the evolution of the in situ reaction, thermodynamic calculations of the reactions proposed by different authors were performed using the HSC Chemistry 6.0 program.

According to the thermodynamic studies of Degang Zhao et al. [66], the total ZrB_2 formation reaction by the interaction between pure Al (99.85% wt.) and salts, at 1173K is (3.1), having the free energy of negative Gibbs formation ($\Delta G = -758.73 \text{ kJ} / \text{mol}$).

$$K_2 ZrF_6 + 2 KBF_4 + 10/3 Al = ZrB_2 + 10/3 AlF_3 + 4 KF$$
(3.1)

For calculations according to the HSC Chemistry 6.0 program, K_2ZrF_6 was considered dissociated into ZrF_4 and 2KF.

$$ZrF_4 + 2 KF + 2 KBF_4 + 10/3 Al = ZrB_2 + 10/3 AlF_3 + 4 KF$$
 (3.2)

In table 3.1. the results of the thermodynamic calculation for this reaction are presented again in (Figure 3.1) Ellingham diagram.

10	Table 5.1. The result of the distinouynamic calculation of the reaction (5.2)												
	$ZrF_4 + 2 KT$	$F + 2 KBF_4 + 10/3$	$3 \text{ Al} = \text{Zr}\text{B}_2 + 1$	$10/3 \text{ AlF}_3 + 4 \text{ KH}_3$	7								
T, °C	deltaH, kJ	deltaS, J/grad	deltaG, kJ	K	Log(K)								
700	-917,920	-272,888	-652,359	1,044E+035	35,019								
720	-918,961	-273,947	-646,891	1,062E+034	34,026								
740	-919,985	-274,967	-641,402	1,178E+033	33,071								
760	-920,990	-275,950	-635,892	1,421E+032	32,153								
780	-921,978	-276,897	-630,364	1,852E+031	31,268								
800	-922,947	-277,809	-624,817	2,600E+030	30,415								
820	-923,898	-278,687	-619,252	3,914E+029	29,593								
840	-924,831	-279,533	-613,669	6,294E+028	28,799								
860	-871,336	-232,198	-608,220	1,095E+028	28,039								
880	-872,128	-232,891	-603,569	2,200E+027	27,342								
900	-872,916	-233,569	-598,905	4,662E+026	26,669								
920	-934,702	-285,798	-593,703	9,858E+025	25,994								
940	-935,477	-286,441	-587,981	2,084E+025	25,319								
960	-936,233	-287,060	-582,245	4,626E+024	24,665								
980	-936,972	-287,654	-576,498	1,077E+024	24,032								

Table 3.1. The result of the thermodynamic calculation of the reaction (3.2)

K - the equilibrium constant of the reaction



Figure 3.1. Ellingham diagram $\Delta G^{o}_{T} = f(T)$ for the formation of molten ZrB₂ (reaction 3.2)





Figure 3.2. Variation of the thermodynamic parameters of the reaction (3.2): a) enthalpy variation; b) entropy variation; c) variation $\ln K = f (1 / T)$

In conclusion, the reaction 3.1. it is thermodynamically possible having a negative ΔG at 1173K (900°C).

CHAPTER 4. Development of AA6063 / ZrB₂ composites by in situ reactions

The literature presents a series of studies on obtaining aluminum matrix composites of series AA7075, A356, AA2024, AA5052, AA2014, AA6061, reinforced with zirconium diboride particles, obtained by aluminothermic reactions at different temperatures (1000 K, 1023 K, 1123 K, 1143 K, 1158 K, 1163 K 1173 K) using different concentrations of KBF₄ (for B) and K_2ZrF_6 (for zirconium) salts.

In Chapter 3 of this doctoral dissertation, determinations were made using the HSC Chemistry 6.0 program regarding the thermodynamics of in situ reactions in order to obtain ZrB_2 particles, for the temperature range 700 - 1000°C.

It was concluded that reaction 3.13 has the highest values in the studied temperature range, for the free energy of Gibbs formation.

$$6KF + 3ZrF_4 + 6KBF_4 + 10AI = 3ZrB_2 + 9KAIF_4 + K_3AIF_6$$
(3.13)

The aluminothermic reaction for the formation of zirconium diboride can be carried out in several stages, the first of which being to obtain aluminium diboride, according to reaction 3.7.

$$6 \text{ KBF}_4 + 9 \text{ Al} = 3 \text{ AlB}_2 + 2 \text{ K}_3 \text{AlF}_6 + 4 \text{ AlF}_3$$
(3.4)

followed by obtaining the zirconium aluminide (ZrAl₃) according to reaction (3.4)

$$6 K_2 ZrF_6 + 13 Al = 3 Al_3 Zr + 2 K_3 AlF_6 + 2 AlF_3$$
(3.7)

și ulterior, în urma reacției dintre AlB₂ și ZrAl₃, cu formarea diborurii de zirconiu, conform reacției (3.8).

$$ZrAl_3 + AlB_2 = ZrB_2 + 4Al \tag{3.8}$$

In the present doctoral thesis we aimed to obtain by aluminothermic reactions, composites with AA6063 alloy matrix reinforced in situ with zirconium diboride particles.

For the study it was desired to obtain composites with different concentrations of reinforcing materials (2.5% ZrB₂, 5% ZrB₂, 7.5% ZrB₂, 10% ZrB₂) at a temperature of 900°C.

4.1. The technological flow of in situ elaboration of composites and the experimental procedure

The in situ technological flow of Al / ZrB_2 composites



Figure 4.1. Schematic of the technological process of in situ elaboration of Al / ZrB_2 composites

4.2. Load calculation

The amounts of K_2ZrF_6 and KBF_4 were calculated according to the reactions below, in order to determine the salt required to obtain boron-containing composites in amounts of 7.5, 15, 22.5, 30 g of ZrB_2 , per 300 g of alloy used according to the general reaction to obtain composite materials (3.13):

$$3K_2ZrF_6 + 6KBF_4 + 10Al = 3ZrB_2 + 9KAlF_4 + K_3AlF_6 +$$
(3.13)

The bars were cast in a preheated steel shell at 200°C (Figure 4.2.)



Figure 4.2. AA6063 / ZrB₂ composite cast bars

Samples were taken from the samples obtained for the characterization from a compositional point of view and the characterization of the physical-mechanical properties. The notations of the samples are: A (2.5% ZrB₂); B (5% ZrB₂); C (7.5% ZrB₂); D (10% ZrB₂).

The samples were processed by metallographic methods (cutting with the DELTA Abrasimet device, bakelite embedding with the SIMPLIMET 1000 device, polishing and sanding with the Buehler Beta / 1 Single device) and subsequently analysed by optical microscopy using the Olympus BX51M Optical Microscope with Olympus U30 camera and Olympus Stream Essentials software. The samples were attacked with Keller reagent (95 ml H₂O, 2.5 ml HNO₃, 1.5 ml HCl and 1 ml HF).

The presence of the present reinforcement phases was confirmed by electron microscopy using the SEM FEI Quanta Inspect F field emission microscope and equipped with an energy dispersion spectrometer (EDS).



A attacked zone 1 x50

A attacked zone 1 x100



A unattacked zone 1 x200 A unattacked zone 1 x500 Figure 4.3. The microstructure of composite A attacked and unattacked on different areas and sizes

The SEM micrograph of the AA6063 matrix and the EDS analysis of the AA6063 / ZrB_2 composite is shown in Figure 4.4.



Figure 4.4. SEM analysis of matrix AA6063 (a), composites AA6063 / ZrB₂ (b, c, d) with 2.5% ZrB₂ and EDS analysis with chemical composition of sample A

The microstructure of the matrix alloy is typically dendritic. The completely dendritic microstructure is absent in the composites, confirming the finishing of the granulation by the fine particles of ZrB_2 and $TiAl_3$ (Figure 4.4.) Which act as granulation finishers. Figure 4.5 shows the microstructures of the attacked and unattacked sample D. In some areas the formation of ZrB2 clusters is observed.



D unattacked zone 1 x200 D unattacked zone 1 x500 Figure 4.5. Microstructure of composite D unattacked and attacked, at different magnifications and in different areas

In Figure 4.6. the SEM analysis of composite D is presented as well as the EDS analysis of Zr-containing particles. The 10% increase in ZrB_2 particles leads, in some areas, to their agglomeration (Figure 4.6 c).



Figure 4.6. ME analysis and EDS analysis with the chemical composition of sample D

XRD analysis of samples at different concentrations of ZrB_2 was performed with the PANalytical X'Pert PRO diffractometer (Figure 4.7 - Figure 4.10) but also with the D8 ADVANCE diffractometer (Figure 4.11).



Figure 4.7. XRD AA6063 / 2,5% ZrB₂





It was highlighted that all samples of composite materials contain ZrB2, Al₃Zr and Al.

The ADVANCE D8 diffractometer is based on the unique platform of the D8 family of diffractometers and is perfectly designed for all X-ray and dispersion diffraction applications, including:

- X-ray diffraction (XRD)

- Pair distribution function (PDF analysis)

- Wide and small angle X-ray scattering (SAXS, WAXS)

D8 ADVANCE has the ability to measure all types of samples, from liquids to free powders, from thin films to solid blocks, on a single instrument [80].









Figure 4.11. XRD analysis of samples A, B, C, D - for the whole range of values of the angle 2θ (a), for ss α Al (j) or for the compounds ZrAl₃ (b, h, i) and ZrB₂ (b, d ÷ k)

For the determinations performed, the databases of the D8 ADVANCE diffractometer were used, namely sheet 00-048-1385 for Al_3Zr and sheet 00-04-0423 for ZrB_2 , in which are specified, for pure substances, the crystallographic systems in which it crystallizes, the parameters elementary cell, interatomic distances, densities, molar masses, etc.

Operator

Seq

16

Operator

y,x,-z

Seq

Quality Mark: Indexed Environment: Ambient Temp: 298.0 K (Assigned by ICDD editor) Status Primary Weight %: Al47.02 Zr52.98 Chemical Formula: Al3 Zr Empirical Formula: Al3 Zr Atomic %: Al75.00 Zr25.00 Compound Name: Aluminum Zirconium Entry Date: 09/01/1998

Radiation: CuKo1 (1.5405 Å) Internal Standard: Si d-Spacing: Guinier Intensity: Visual

SPGR: I4/mmm (139) Crystal System: Tetragonal Volume: 277.74 Å³ Author's Cell [a: 4.009 Å **c:** 17.281 Å **Z:** 4.00 MolVol: 69.44 c/a: 4.311] Calculated Density: 4.117 g/cm³ **SS/FOM:** F(24) = 25.6(0.022, 43)

Space Group: I4/mmm (139) Molecular Weight: 172.16 g/mol Crystal Data [XtlCell a: 4.009 Å XtlCell b: 4.009 Å XtlCell c: 17.281 Å XtlCell α: 90.00° XtlCell β: 90.00° XtlCell y: 90.00° XtlCell Vol: 277.74 Å³ XtlCell Z: 4.00 c/a: 4.311 a/b: 1.000 c/b: 4.311] Reduced Cell [RedCell a: 4.009 Å RedCell b: 4.009 Å RedCell c: 9.094 Å RedCell a: 102.73° RedCell γ: 90.00° RedCell β: 102.73° RedCell Vol: 138.87 Å³]

Atomic parameters are cross-referenced from PDF entry 04-001-2571 Space Group Symmetry Operators: Operator Operator Operator Seq Seq Seq Seq Operator 10 y,-x,-z 1 x,y,z 4 x,y,-z 7 x,-y,z

1 2 3	x,y,z -x,-y,-z -x,-y,z	4 5 6	x,y,-z -x,y,z x,-y,-z	7 8 9	Х,- -Х, -У,	y,z y,-z x,z	10 11 12	y,-x,-z y,-x,z -y,x,-z	:	13 14 15	y,x,z -y,-x,-z -y,-x,z	10
Atomi	c Coordi	nates:										
Atom	Num	Wyckoff	Symmetry	х	y	z	SOF	IDP	AET			
Zr	1	4e	4mm	0.0	0.0	0.11886	1.0		12-b			
AI	2	4e	4mm	0.0	0.0	0.37498	1.0		12-b			
AI	3	4d	-4m2	0.0	0.5	0.25	1.0		12-b			
AI	4	4c	mmm.	0.0	0.5	0.0	1.0		12-b			
_												

Crystal (Symmetry Allowed): Centrosymmetric

Subfiles: Inorganic, Metal & Alloy Pearson Symbol: tI16.00 Prototype Structure [Formula Order]: Zr Al3 Prototype Structure [Alpha Order]: Al3 Zr LPF Prototype Structure [Formula Order]: Zr Al3,tI16,139 LPF Prototype Structure [Alpha Order]: Al3 Zr,tI16,139

00-002-1093 (Deleted), ✓ 04-001-2571 (Alternate), ✓ 04-001-3330 (Alternate), ✓ 04-001-3351 Cross-Ref PDF #'s: (Alternate), ✓ 04-001-3544 (Alternate), ✓ 04-003-4015 (Primary), ✓ 04-003-4563 (Alternate), ✓ 04-004-7164 (Alternate), ✓ 04-008-8213 (Alternate), ✓ 04-010-6142 (Alternate)

References:		
Туре	DOI	Reference
Primary Reference		Clark, N., School of Physical Sciences, The Flinders Univ. of South Australia, Adelaide, Australia. Private Communication (1996).
Crystal Structure		Crystal Structure Source: LPF.
Powder Data (Additonal References)		Clark, N., Wu, E. J. Less-Common Met. 163, 227 (1990).
Structure		Ma, Y., Romming, C., Lebech, B., Gjonnes, J., Tafto, J. Acta Crystallogr., Sec. B: Struct. Sci. 48, 11 (1992).
Unit Cell		Brauer, G. Z. Anorg. Chem. 242, 15 (1939).

Database Comments: Additional Patterns: To replace 00-002-1093. Sample Preparation: Synthesized from elements in an argon arc furnace. Warning: Lines with abs(delta 2Theta)>0.06 DEG.

d-Spacing	d-Spacings (24) - Al3 Zr - 00-048-1385 (Stick, Fixed Slit Intensity) - Cu Ka1 1.54056 Å														
<u>2θ (°)</u>	d (Å)	I	h	k	1	*	<u>2θ (°)</u>	d (Å)	I	h	k	1	*		
20.54216	4.320000	50	0	0	4		53.44467	1.713000	5	2	1	3			
22.72346	3.910000	60	1	0	1		57.87355	1.592000	10	2	1	5			
27.08041	3.290000	20	1	0	3		63.24989	1.469000	60	2	0	8			
31.51994	2.836000	30	1	1	0		65.85797	1.417000	50	2	2	0			
34.22209	2.618000	20	1	0	5		69.69922	1.348000	5	2	2	4			
37.93268	2.370000	100	1	1	4		70.66048	1.332000	5	3	0	1			
41,76420	2.161000	50	0	0	8		72.03042	1.310000	10	2	1	9			
45.20943	2.004000	60	2	0	Ō		73,72641	1.284000	50	1	1	12			
50,10689	1.819000	10	2	0	4		74.81450	1.268000	5	3	1	0			
51.16005	1.784000	10	2	1	1		75.23173	1.262000	5	1	0	13			
52.77986	1.733000	5	1	Ó	9		78.61082	1.216000	60	3	1	4			
53.21001	1.720000	5	1	1	8		81.08702	1.185000	50	2	2	8			

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Figure 4.12. Sheet 00-048-1385 for Al₃Zr

- -

Status PrimaryQuality Mark: StarEnvironment: AmbientTemp: 298.0 K (Assigned by ICDD editor)Chemical Formula: Zr B2Empirical Formula: B2 ZrWeight %: B19.16 Zr80.84Atomic %: B66.67 Zr33.33Compound Name: Boron ZirconiumAlternate Name: zirconium diborideCAS Number: 12045-64-6Entry Date: 09/01/1984

Radiation: CuKo1 (1.5406 Å) Internal Standard: Si d-Spacing: Diffractometer Cutoff: 22.10 Å Intensity: Diffractometer - Peak

 Crystal System:
 Hexagonal
 SPGR:
 P6/mmm (191)

 Author's Cell [a: 3.16870(8) Å
 c: 3.53002(10) Å
 Volume: 30.70 Å³
 Z: 1.00
 MolVol: 30.70
 c/a: 1.114]

 Calculated Density:
 6.104 g/cm³
 Color:
 Dark gray
 SS/FOM: F(23) = 179.7(0.0056, 23)
 c/a: 1.114]

 Space Group: P6/mmm (191)
 Molecular Weight: 112.84 g/mol

 Crystal Data [XtiCell a: 3.169 Å
 XtiCell b: 3.169 Å
 XtiCell c: 3.530 Å
 XtiCell a: 90.00°
 XtiCell β: 90.00°

 XtiCell y: 120.00°
 XtiCell Vol: 30.70 Å³
 XtiCell Z: 1.00
 c/a: 1.114
 a/b: 1.000
 c/b: 1.114]

 Reduced Cell [RedCell a: 3.169 Å
 RedCell b: 3.169 Å
 RedCell c: 3.530 Å
 RedCell a: 90.00°

 RedCell β: 90.00°
 RedCell p: 120.00°
 RedCell b: 3.169 Å
 RedCell c: 3.530 Å
 RedCell a: 90.00°

Atomic parameters are cross-referenced from PDF entry 04-004-2991 Space Group Symmetry Operators:

Seq	Operato	or Seq	Operator	Seq	Operator	Se	q Op	erator	Seq	Operator	Seq	Operator
1 2 3 4	x,y,z -x,-y,-z -y,x-y,z y,-x+y,-z	5 6 7 8	-x+y,-x,z x-y,x,-z y,x,z -y,-x,-z	9 10 11 12	-x,-x+y,z x,x-y,-z x-y,-y,z -x+y,y,-z	13 14 15 16	-x,- x,y y,-> -y,>	-y,z ;,-z x+y,z x-y,-z	17 18 19 20	x-y,x,z -x+y,-x,-z -y,-x,z y,x,-z	21 22 23 24	x,x-y,z -x,-x+y,-z -x+y,y,z x-y,-y,-z
Atom	ic Coorai	nates:										
<u>Atom</u>	Num	Wyckoff	Symmetry	x	v	z	SOF	IDP	AET			
Zr B	1 2	1a 2d	6/mmm -6m2	0.0 0.33333	0.0 0.66666	0.0 0.5	1.0 1.0		20-a 3#b			

Crystal (Symmetry Allowed): Centrosymmetric

 Subfiles: Common Phase, Forensic, Inorganic, Metal & Alloy, NBS Pattern
 Pearson Symbol: hP3.00

 Prototype Structure [Formula Order]: Al B2
 Prototype Structure [Alpha Order]: Al B2

 LPF Prototype Structure [Formula Order]: Al B2, hP3, 191
 LPF Prototype Structure [Alpha Order]: Al B2, hP3, 191

 ANX: NO2
 NO2

Cross-Ref PDF #'s	00-006-0610 (Deleted), < 04-001-1203 (Primary), < 04-001-3237 (Alternate), < 04-002-0014 (Alternate), < 04-002-1160 (Alternate), < 04-003-5557 (Alternate), < 04-003-6094 (Alternate), < : 04-003-6177 (Alternate), < 04-003-6284 (Alternate), < 04-004-2991 (Alternate), < 04-004-5887 (Alternate), < 04-004-7147 (Alternate), < 04-004-7151 (Alternate), < 04-006-0200 (Alternate), < 04-006-2031 (Alternate), < 04-006-2211 (Alternate)
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References:		
Туре	DOI	Reference
Primary Reference Additional Pattern Additional Reference Crystal Structure		Natl. Bur. Stand. (U. S.) Monogr. 25 20, 113 (1984). Kiessling, R. Acta Chem. Scand. 3, 90 (1949). Norton, J. et al. Metall. Trans. 185, 749 (1949). Crystal Structure Source: LPF.

Additional Patterns: To replace 00-006-0610. ANX: NO2. Sample Preparation: The sample was prepared at NBS, Gaithersburg, Maryland, USA. Structures: Kiessling (1949) and Norton et al. (1949) studied the structure of "Zr B2". Unit Cell Data Source: Powder Diffraction.

d-Spacings	d-Spacings (23) - Zr B2 - 00-034-0423 (Stick, Fixed Slit Intensity) - Cu Ka1 1.54056 Å														
<u>20 (°)</u>	d (Å)	I	h	k	1	*	<u>20 (°)</u>	d (Å)	I	h	k	1	*		
25.20438	3.530470	30	0	0	1		90.84991	1.081350	10	1	0	3			
32.60018	2.744450	67	1	0	0		95.91463	1.037210	7	2	1	0			
41.65686	2.166320	100	1	0	1		101.43860	0.995125	14	2	1	1			
51.74466	1.765210	9	0	0	2		109.26193	0.944618	4	1	1	3			
58.18250	1.584280	19	1	1	0		114.72865	0.914695	3	3	0	0			
62.51314	1.484530	21	1	0	2		118.95322	0.894196	10	2	1	2			
64.40219	1.445470	16	1	1	1		119.18013	0.893155	6	2	0	3			
68.29627	1.372230	8	2	0	0		120.89406	0.885481	4	3	0	1			
74.06876	1.278910	16	2	0	1		121.58804	0.882467	2	0	0	4			
81.58260	1.179050	13	1	1	2		132.93134	0.840164	4	1	0	4			
81,76758	1.176850	1	0	0	3		143.03853	0.812163	7	3	0	2			
90.65145	1.083200	8	2	0	2										

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Figure 4.13. Sheet 00-034-0423 for ZrB₂

Extraction of ZrB₂ particles from the composite materials was performed by dissolving in 35% HCl solution followed by filtration and drying.

SEM analysis of ZrB_2 powders was performed with the SEM plant (VEGA II LMU). The results are given in Figure 4.14.





(o) Figure 4.14. SEM analysis of ZrB₂ powders extracted from composites

ZrB₂ powders extracted from the samples of cast composites with dimensions between 5µm and 20µm were performed HRTEM analysis of the extracted powders (Figure 4.15).







CHAPTER 5. Physio-mechanical properties of AA6063 / ZrB₂ composites

The mechanical properties followed and determined for the elaborated composites were: hardness, mechanical resistance to breakage, tensile, mechanical resistance to compression. Important properties of the reinforcement elements obtained in situ are presented in Table 5.1.

IUPAC Name	Theoretical chemical formula, [CASRN]	Crystallographic system, network parameters. Pearson symbol, space group, the type of structure, Z	Thermal conductivity (k.Wm ⁻¹ K ⁻¹)	Specific thermal capacity (c _p .J kg ⁻¹ K ⁻	Coefficient of linear thermal expansion (α, 10 ⁻⁶ K ⁻¹)
Zirconium diboride	ZrB ₂ [12045- 64-6] 112.846	Hexagonal a = 0,3169 nm c = 0,3530 nm $hP3, P6/mmm, AlB_2$ type (Z = 1)	57,9	392,54	5,5-8,3

Table 5.1. Physical properties of zirconium diboride (ZrB₂) [81]

5.1. Hardness

The hardness was determined using a microdurimeter produced by Leco M-400-G, year of manufacture 2004.



Sample B x200

Sample B x200



Figure 5.1. Determination of the size of the cavities left after the tests for the determination of Vickers microhardness by measurement using optical microscopy

5.2. Tensile strength





Figure 5.2. Samples used to determine tensile and elongation strengths

Data for tensile and elongation strength were taken from the results obtained using the Instron Universal Testing Machine 8872 at room temperature using cylindrical samples 15 mm long and 5 mm in diameter. On average, three samples were used for each test.





Figure 5.3. Tensile test results for manufactured composite materials



5.3. Compression strength



Figure 5.4. Compression test results for manufactured composites

5.4. Study of fractography of samples of cast composites

The in-depth study of the fracture surfaces was performed by SEM analysis using the HITACHI HD-2300 microscope (scanning transmission electron microscope).



Figure 5.5. SEM in fracture at composite AA6063/2,5% ZrB₂ (Sample A)



Figure 5.6. SEM in fracture at composite AA6063/5% ZrB₂ (Sample B)



Figure 5.7. SEM in fracture at composite AA6063/7,5% ZrB₂ (Sample C)



Figure 5.8. SEM rupture at composite AA6063/10% ZrB₂ (Sample D)

5.5. The coefficient of expansion and diffusivity of the elaborated composites

Any property of a material studied in a temperature range can be considered a thermophysical property. However, traditionally, thermal expansion, thermal conductivity and thermal diffusivity are considered to be the most common fundamental thermophysical properties.

A. Thermal expansion

The purpose of this method is to determine the coefficient of linear thermal expansion (CTE) for some materials with coefficients of linear thermal expansion greater than ± 1 micron / ° C, using a dilatometer with high purity sintered alumina components.

Thermal expansion coefficient analysis is a thermal method of tracking the dimensional variability of a sample as a function of temperature and time.

The determination of the thermal expansion coefficient was performed with the help of the thermal dilatometer model Unitherm 1161V (Fig. 5.9).



Figure 5.9. Vertical thermal dilatometer, model Unitherm 1161V

The temperature values between which this test is performed are in the range of $20 - 600^{\circ}$ C. This method is not used for materials with coefficients of thermal expansion less than $1 \times 10^{-6} / {^{\circ}}$ C; below this value the interferometer or other capacitive measuring techniques are recommended.

The experimental data, respectively the dilatograms, for the four samples, taken from the composite materials with 2.5%, 5.0%, 7.5% and 10.0% ZrB2, respectively, are presented in Fig. 5.10.





Figure 5.10. Dilatograms, for the four samples, taken from composite materials with (a) 2.5%, (b) 5.0%, (c) 7.5% and (d) 10.0% ZrB₂, respectively

In Figure 5.11. the results are presented in numerical and graphical format, respectively, of the percentage coefficient of thermal expansion (E%), of the instantaneous coefficient of thermal expansion (A-CTE) and respectively of the average coefficient of linear thermal expansion (A-CTE).





Figure 5.11. The results in graphical and numerical format, respectively, of the percentage coefficient of thermal expansion (E%), of the instantaneous coefficient of thermal expansion (CTE) and of the average coefficient of linear thermal expansion (A-CTE), respectively, for: (a) 2.5%ZrB₂; (b) 5.0%ZrB₂; (c) 7.5%ZrB₂; (d) 10.0%ZrB₂

B) Thermal diffusivity

Thermal diffusivity is one of the most important transient thermal properties of materials. Since thermal diffusivity is a measure of the speed with which heat passes through a material, its importance is indirect but seems linearly related to the speed of things around us.

The experimental diffusivity data, for the four samples, taken from composite materials with 2.5%, 5.0%, 7.5% and 10.0% ZrB₂, respectively, are presented in Tables 5.2, 5.3, 5.4 and 5.5.

Segment	Temperatura	Amediu	a 1	02	0.3	0.4	0.5	Q.6
	[°C]	[cm ² /sec]						
А	89	0,6993	0,7062	0,6946	0,7113	0,7357	0,5838	0,7643
В	209	0,7671	0,7629	0,7713	0,7552	0,7773	0,7651	0,7708
С	329	0,7613	0,7616	0,7626	0,7598	0,7615	0,7603	0,7621
D	423	0,7362	eroare	eroare	0,7364	eroare	eroare	0,7401

Table 5.2 – Thermal diffusivity values for the composite sample with 2.5% ZrB₂

E	529	0,6888	0,7106	0,7125	0,7242	eroare	0,6609	0,6359

Segment	Temperature	Amediu	α1	α2	Q 3	α4	α5	A 6
	[°C]	[cm ² /sec]						
А	89	0,6999	0,6966	0,7033	0,6965	0,6967	0,7033	0,7032
В	239	0.7725	0,7698	0,7767	0,7750	0,7712	0,7715	0,7710
С	369	0,6899	0,6898	0,6899	0,6897	0,6999	0,6900	eroare
D	427	0,6558	0,6490	0,6558	0,6507	0,6609	0,6558	0,6626
Е	485	0,5945	0,5943	0,5842	0,5945	0,6008	0,5944	0,5990

Table 5.3 – Thermal diffusivity values for the composite sample with 5% ZrB₂

Table 5.4 – Thermal diffusivity values for the composite sample with 7.5% ZrB_2

Segment	Temperature	Amediu	α1	a 2	Q 3	α4	α5	a 6
	[°C]	[cm ² /sec]						
А	75	0,7776	0,7776	0,7777	0,7778	0,7776	0,7775	0,7778
В	236	0,7509	0,7446	0,7456	0,7539	0,7549	0,7560	0,7508
С	388	0,7063	0,7146	0,6981	0,7063	eroare	0,7062	eroare
D	465	0,6550	0,6649	0,6650	0,6651	0,6650	0,6648	0,6652
Е	531	0,5843	eroare	0,5842	0,5844	0,5843	eroare	eroare

Table 5.5 – Thermal diffusivity values for the composite sample with 10% ZrB_2

Segment	Temperature	Amediu	α1	a 2	Q 3	Q 4	α5	A 6
	[°C]	[cm ² /sec]						
А	112	0,7342	0,7187	0,7446	eroare	0,7187	0,7446	0,7446
В	254	0.7728	eroare	0,7725	0,7765	0,7765	0,7693	0,7693
C	398	0,7374	0,7374	0,7417	0,7291	eroare	0,7374	0,7417
D	475	0,6570	eroare	eroare	0,6273	0,6867	0,6867	0,6273
Е	530	0,6476	0,6364	0,6365	0,6589	0,6570	0,6484	0,6485

The graphical representation of the thermal diffusivity as a function of temperature, corresponding to the values presented in Tables 5.6, 5.7, 5.8 and 5.9, is shown in Fig. 5.12.



 $(a) - 2.5\% ZrB_2$



Figure 5.12 - Graphical representation of thermal diffusivity as a function of temperature, corresponding to the values presented in Tables 5.1 (a), 5.2 (b), 5.3 (c) and 5.4 (d).

In order to verify the correctness of the obtained results, a reference material, respectively a composite graphite (Thermal Graphite-TG) was tested; the tabulated values of the thermal diffusivity, as well as the graphical representation, are presented in Table 5.6, respectively Fig. 5.13.

Segment	Temperature	Amediu	α1	a 2	Q 3	Q 4	α5	a 6
	[°C]	[cm ² /sec]						
А	90	0,4300	0,4277	0,4212	0,4236	0,4558	0,4355	0,4164
TG(etalon)	100	0,4440						
В	210	0,3272	0,3283	0,3282	0,3222	0,3306	0,3250	0,3286
TG(etalon)	200	0,3540						
С	329	0,2600	0,2664	0,2604	0,2604	0,2600	0,2553	0,2573
TG(etalon)	300	0,2880						
D	424	0,2256	0,2321	0,2257	0,2254	0,2251	0,2223	0,2229
TG(etalon)	400	0,2430						
Е	529	0,1973	0,1980	0,1986	0,1983	0,1979	0,1975	0,1936
TG(etalon)	500	0,212						

Table 5.6 - Thermal diffusivity values for the standard composite graphite sample (**Thermal Graphite-TG**), measured and compared with the values in the Compliance Bulletin



Figure 5.13 - Graphical representation of thermal diffusivity as a function of temperature, corresponding to the values presented in Table 5.5

CHAPTER 6. Summary of the main scientific and technical contributions of the author

Increasing demand for high-density, low-density materials in the aerospace and automotive industries makes aluminium matrix composites (AMCS) a leading candidate for a number of applications.

Thus, a large number of AMCS composites replace conventional aluminium alloys due to the combination of properties such as: high wear resistance, low coefficient of thermal expansion and a high strength / mass ratio. Graphite particles, carbides (SiC), oxides (Al_2O_3), nitrides are mainly used as reinforcement materials.

From different ceramic reinforcement materials, we chose zirconium boride (ZrB_2) because it has high melting temperature, high hardness, high density ($2g / cm^3$), high coefficient of thermal expansion ($6.88 \times 10^{-6} K^{-1}$), thermal inertia, high electrical conductivity, high chemical inertia.

AMCS composites can be produced in either solid or liquid state. From the two categories of processes, we chose the liquid process. This procedure can be "ex-situ" or "in-situ".

For the production of AA6063 / ZrB_2 composites we chose the in-situ process. While the "ex-situ" process involves mixing the particles in the melt, the "in-situ" process consists of chemical reactions between elements or between elements or compounds to obtain the reinforcing particles directly in the melt.

The merit of the "in-situ" process of forming ZrB_2 in the aluminium alloy melt is that it generates fine ceramic particles of ZrB_2 and Al_3Zr , a very good interfacial bond between the matrix and the particles formed by the reaction between K_2ZrF_6 , KBF₄ and Al, and the ZrB_2 particles increase the ductility when finishing the granulation of the cast products.

CHAPTER 7. Conclusions and further research directions

The present work led to the realization of new materials with controllable properties in the radial direction. The following results of the thesis can be considered as original:

• Extensive documentary study on composite materials - their classification according to the basic matrix and the reinforcing elements.

• Documentary study on the structure and properties of composite materials compared to the structure and properties of classical metallic materials.

• The study of the thermodynamics of the phenomena that take place in the system AA6063 - K_2ZrF_6 - KBF₄ - Na₃AlF₆ during the aluminothermic reaction, at different concentrations.

 \bullet Microstructural characterization of AA6063 / ZrB_2 composites, by optical and electron microscopy (SEM, TEM and HRTEM).

• Characterization of composites obtained in-situ by X-ray diffraction (XRD) and energy dispersive spectroscopy (EDS) for different phases formed.

 \bullet TEM analysis of the $\rm ZrB_2$ compound to examine the structure, composition, and properties in detail.

 \bullet Vickers microhardness in different areas of composite materials reinforced with ZrB_2 ceramic particles.

• The influence of ZrB₂ on the breaking of composites.

• Analysis of the granulometric distribution as a function of volume for different concentrations of reinforcing particles, resulting from the reactions in the system AA6063 - K_2ZrF_6 - KBF₄ - Na₃AlF₆.

• Vickers microhardness in different areas of functional gradient materials reinforced with ZrB₂ ceramic particles.

• Comparative analysis of expansion coefficients according to the concentrations of reinforcing elements obtained in-situ.

 \bullet Comparative analysis of the diffusivity of composite materials AA6063 / ZrB_2, depending on the concentrations of reinforcing elements obtained in-situ.

Future directions of research

1. Analysis of the physical-mechanical behaviour of reinforcing composites with different borons (ZrB₂, TiB₂, VB₂, etc.) and completion of mechanical test tests.

2. Determination and mathematical modelling of other correlations such as tensile strength or modulus of elasticity and some material constants.

3. Obtaining functional gradient materials using AA6063 / $\rm ZrB_2$ composites as starting materials.

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